



## Week 14: Beam Buckling

- Euler Buckling
- Effective length for buckling
- Effect of eccentricity

# Stability- different criteria for resisting loads



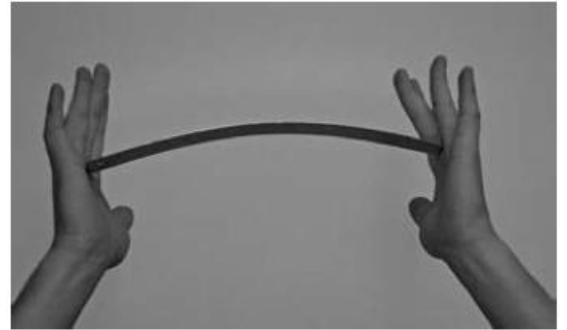
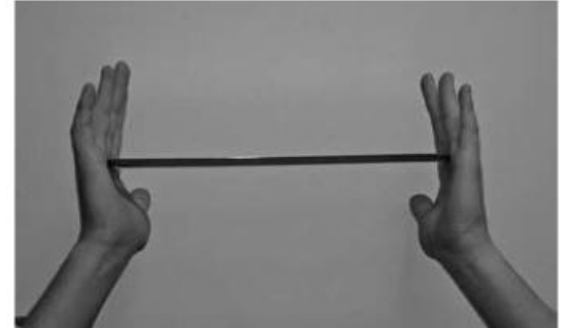
Strength: the ability of a structure to withstand a load without the development of excessive stress



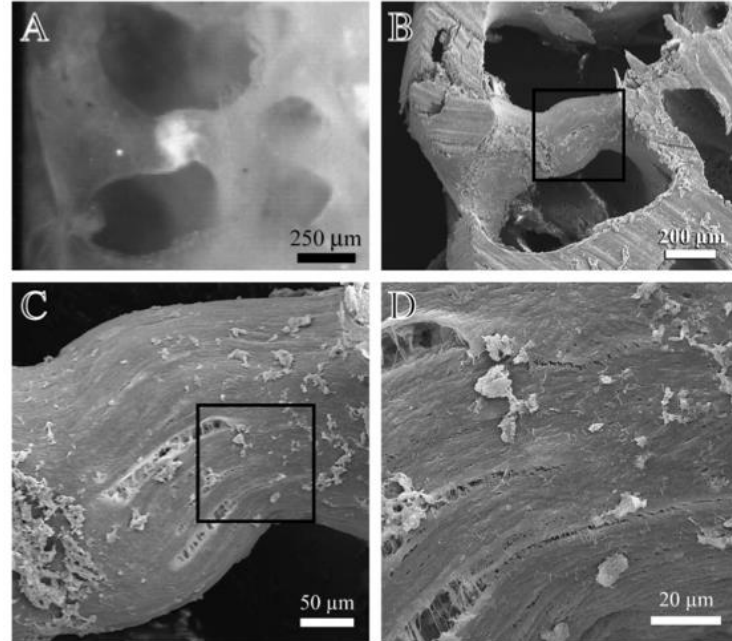
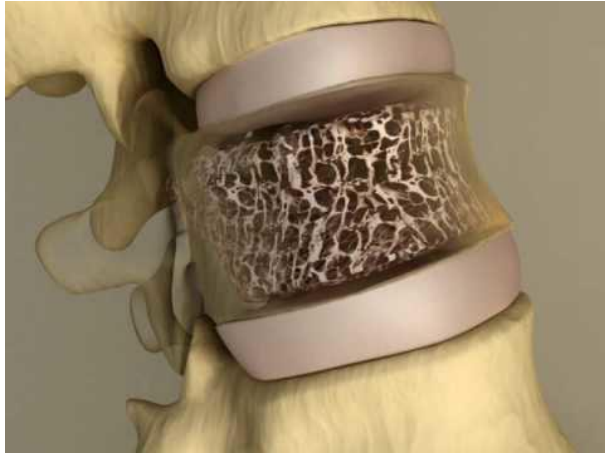
Stiffness: the ability of a structure to withstand a load without developing excessive deformation.

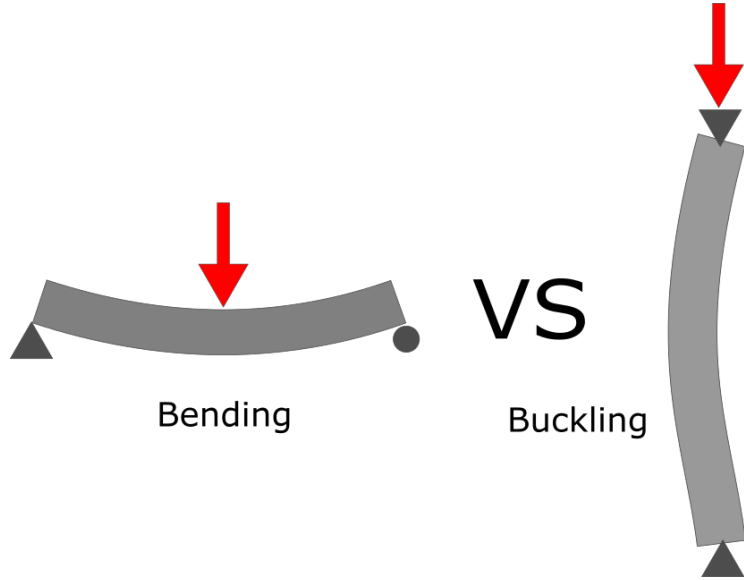


Stability: the ability of a structure to withstand a load without experiencing a sudden change in configuration



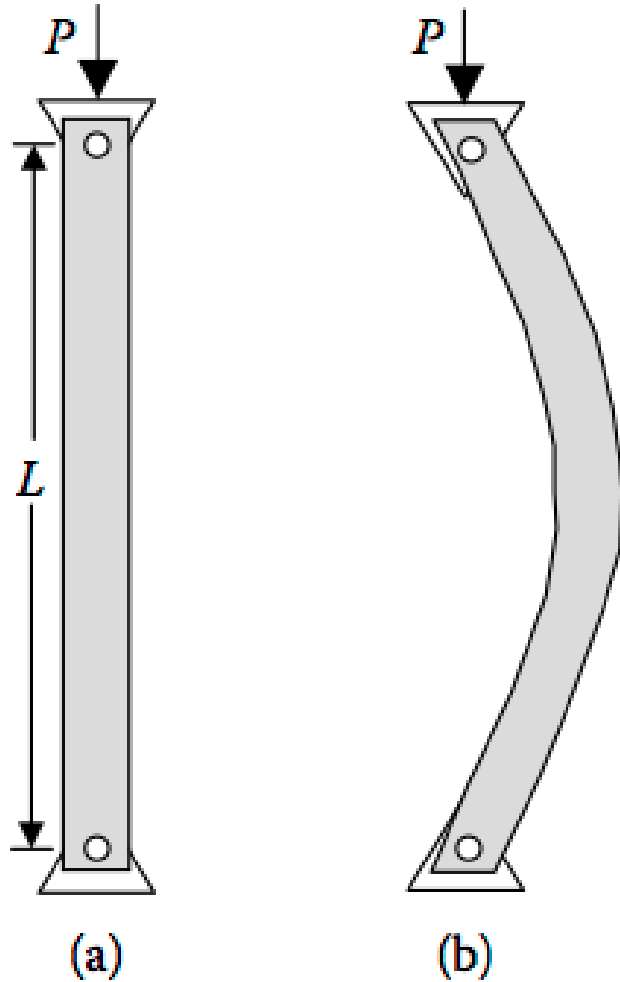
# Buckling is important from the macro to the microscale





# Buckling

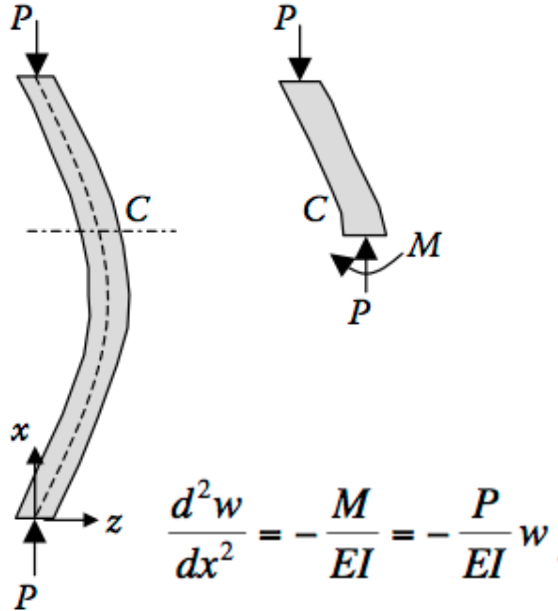
- Buckling is a type of instability that occurs when a beam fails under a compressive load much smaller than the load necessary to reach the yield stress
- In buckling, the failure occurs because the applied load results in a sudden deformation in a perpendicular direction.



# Euler Buckling

Two regimes of deformation when a beam is loaded in compression:

1. If the axial load on a beam is small, the change in length will be due to compressive strain.  $P < P_{cr}$
2. If the axial load on a beam  $P$  is larger than the critical load  $P_{cr}$ , then the beam becomes unstable and a small perturbation will result in buckling of the beam.

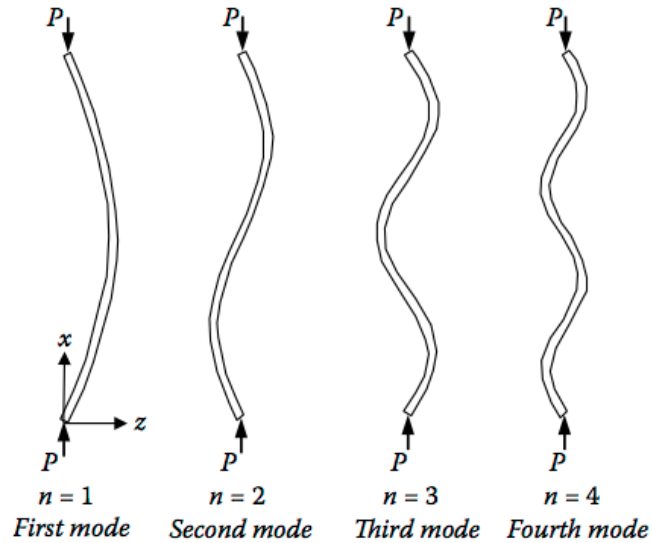


$$\frac{d^2 w}{dx^2} = -\frac{M}{EI} = -\frac{P}{EI} w$$

$$\frac{d^2 w}{dx^2} + \frac{P}{EI} w = 0$$

## Euler Buckling

We can derive the Euler Buckling formula using the method of sections through the buckled beam.



## Euler Buckling

The differential equation has multiple solutions:

This results in multiple buckling modes:

$$P = \frac{n^2 \pi^2 EI}{L^2}$$

- For the critical buckling load we get then *Euler's Buckling Formula*:

$$P_{cr} = \frac{\pi^2 EI}{L^2} .$$

- With the shape of the buckled beam:

$$w(x) = a \sin\left(\frac{\pi}{L} x\right)$$

- The second moment of area (I) should be taken around the axis around which the beam buckles. This is in general the axis with the smallest second moment of area.

$$P_{cr} = \frac{\pi^2 EI}{L_e^2}$$

### Effective Length $L_e$ as Function of Supports

End Conditions	Effective Length
Fixed-Free	$L_e = 2L$
Pinned-Pinned	$L_e = L$
Fixed-Pinned	$L_e = 0.7L$
Fixed-Fixed	$L_e = 0.5L$

## Euler Buckling : Effective length

- The Euler Formula we have derived here only deals with a beam with pinned supports on each end.
- The type of support however greatly influences the critical load and the buckling behavior.
- Euler's formula can be extended towards other types of support by using the concept of the effective length.

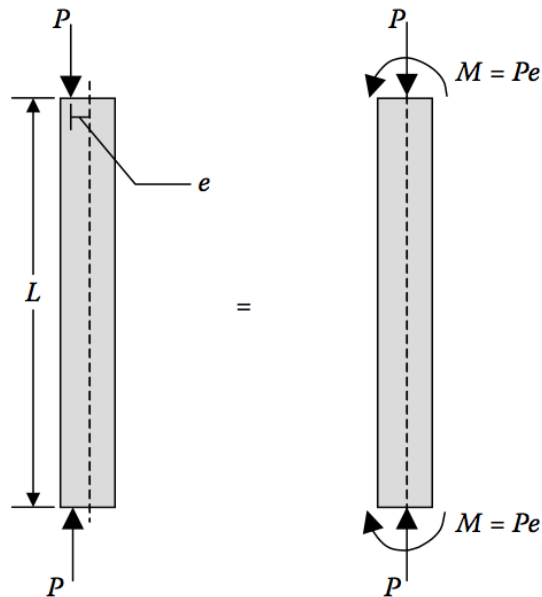
- Critical buckling stress:

$$\sigma_{cr} = \frac{P_{cr}}{A} = \frac{\pi^2 EI}{L_e^2 A}$$

- Using the definition of the radius of gyration  $r = \sqrt{\frac{I}{A}}$ :

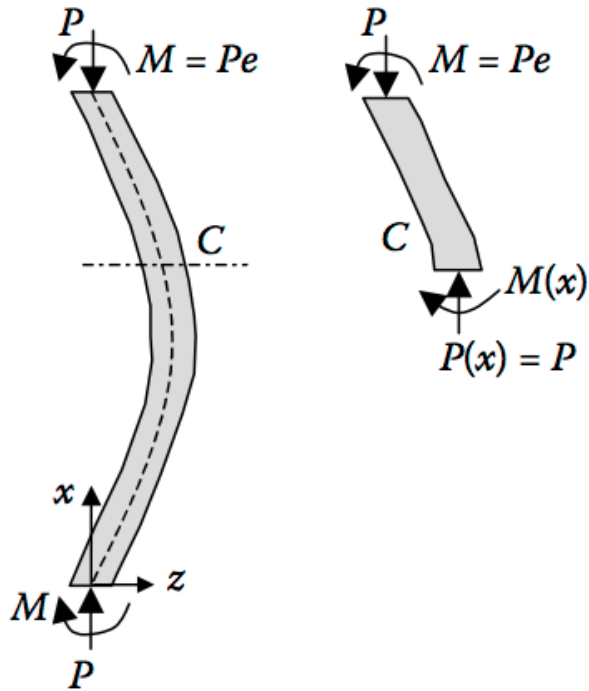
$$\sigma_{cr} = \frac{\pi^2 E A r^2}{L_e^2 A} = \frac{\pi^2 E r^2}{L_e^2} = \frac{\pi^2 E}{(L_e / r)^2}$$

- $L_e/r$  is the slenderness ratio



## Buckling: Effect of eccentricity

- So far, we've looked at beams that were loaded through the centroid of the column
- Often, the load is offset from this axis: eccentric loading
- We calculate the behavior of a beam pinned at both ends with eccentric load.
- We can model the eccentricity with an axial load and a moment at the supports



## Buckling: Effect of eccentricity

- Centric load:  $P$
- Moment:  $M = P \cdot e$
- This means that the beam bends even under small loads without the beam buckling
- We solve the now inhomogeneous differential equation:

$$\frac{d^2 w}{dx^2} + \frac{P}{EI} w = -\frac{P}{EI} \cdot e$$

- For the deflection we then get:

$$w(x) = e \left\{ \tan \left( \sqrt{\frac{P}{EI}} \cdot \frac{L}{2} \right) \cdot \sin \left( \sqrt{\frac{P}{EI}} \cdot x \right) + \cos \left( \sqrt{\frac{P}{EI}} \cdot x \right) - 1 \right\}$$

- And for the maximum deflection:

$$w_{max} = w(L/2) = e \left[ \sec \left( \sqrt{\frac{P}{EI}} \frac{L}{2} \right) - 1 \right]$$

- The critical buckling load is then

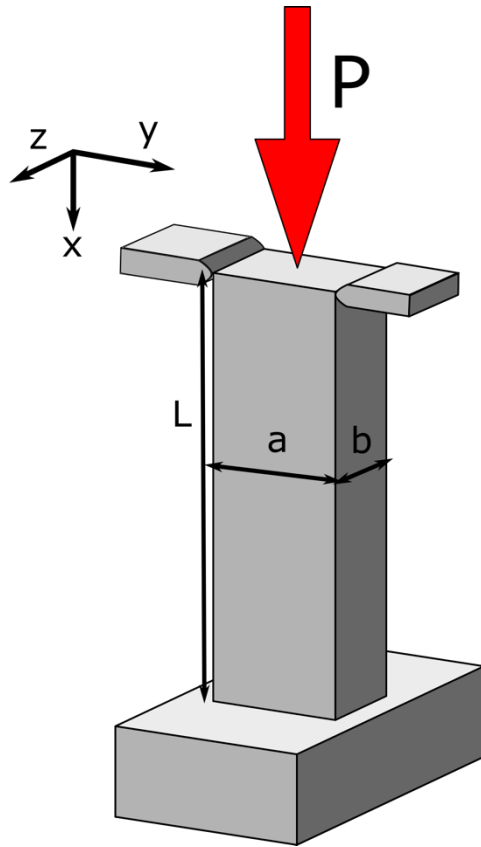
$$P_{cr} = \frac{\pi^2 EI}{L^2}$$

# Buckling: Effect of eccentricity

- Maximum stress in the beam is given by the compressive stress and the bending stress:

$$\sigma_{max} = -\frac{P}{A} \left[ 1 - \frac{ec}{r^2} \sec \left( \sqrt{\frac{P}{EI}} \frac{L}{2} \right) \right]$$

$$\sigma_{max} = -\frac{P}{A} \left[ 1 - \frac{ec}{r^2} \sec \left( \sqrt{\frac{P}{P_{cr}}} \frac{\pi}{2} \right) \right]$$



## Example Buckling

An aluminum column of length  $L$  and rectangular cross section has a fixed end  $B$  and supports a centric axial load at  $A$ .

Two smooth and rounded fixed plates restrain end  $A$  from moving in one of the vertical planes of symmetry but allow it to move in the other plane.

- Determine the ratio  $a/b$  of the two sides of the cross section corresponding to the most efficient design against buckling.
- Design the most efficient cross section for the column, knowing that  $L = 50$  cm,  $E = 70$  GPa,  $P = 22$  kN and that a safety factor of 2.5 is required.